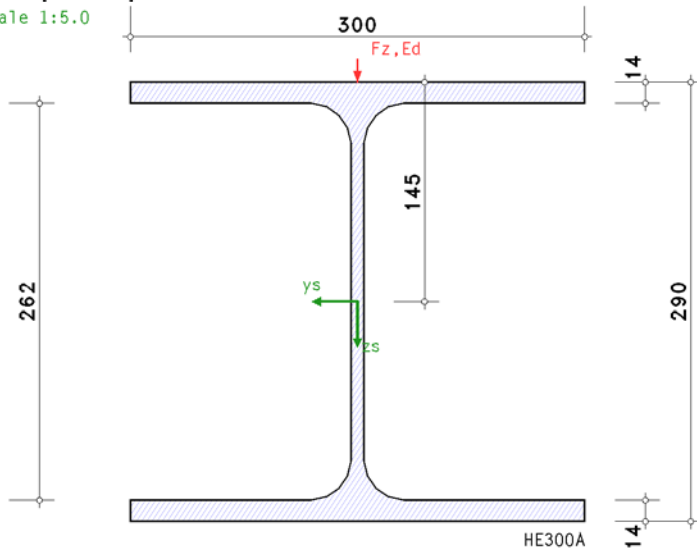


1. input report

scale 1:5.0



steel grade

steel grade S235

cross-section

beam: section HE300A

loading

internal forces and moments at limit state of resistance (ULS):

Lk 1: beam 2

$$M_{y,Ed} = -264.0 \text{ kNm}, V_{z,Ed} = 187.0 \text{ kN}$$

Lk 2: beam 1

$$M_{y,Ed} = -351.0 \text{ kNm}, V_{z,Ed} = 75.0 \text{ kN}$$

transverse loading on top flange:

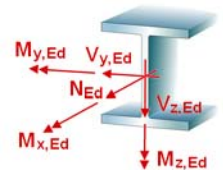
vertical single load $F_{z,Ed,ULS} = 180.00 \text{ kN}$, rigid loading length $s_s = 300.0 \text{ mm}$

verification in beam field

partial safety factors for material

resistance of cross-sections $\gamma_{M0} = 1.00$

resistance of members in stability failure $\gamma_{M1} = 1.10$



2. verification der local loading

assumption: flange induced web buckling is excluded.

assumption: plated structures-/shear buckling is excluded.

cross-sectional properties: $A = 112.53 \text{ cm}^2$, $z_s = 145.0 \text{ mm}$, $I_y = 18263.71 \text{ cm}^4$, $y_s = -0.0 \text{ mm}$, $I_z = 6309.56 \text{ cm}^4$

effective loading length $l_{eff} = s_s + 2 \cdot t_f = 328.0 \text{ mm}$

referring to outer edge of flange $s_s = l_{eff} - 2 \cdot t_f = 300.0 \text{ mm}$ / auf den webanschnitt $s_w = l_{eff} + 2 \cdot r = 382.0 \text{ mm}$

2.1. compression of web (ULS)

permissible stresses: $\sigma_{Rd} = f_y / \gamma_{M0} = 235.0 \text{ N/mm}^2$, $\tau_{Rd} = f_y / (3^{1/2} \cdot \gamma_{M0}) = 135.7 \text{ N/mm}^2$

compression of single load at first cut of web:

local stresses $\sigma_{\alpha z,Ed} = -55.4 \text{ N/mm}^2$, $\tau_{\alpha z,Ed} = 0.0 \text{ N/mm}^2$

$|\sigma_{\alpha z,Ed}| = 55.4 \text{ N/mm}^2 < \sigma_{Rd} = 235.0 \text{ N/mm}^2 \Rightarrow U = 0.236 < 1$ **ok**

stresses at first cut of web:

Lk 1: $M_{y,Ed} = -264.0 \text{ kNm}$, $V_{z,Ed} = 187.0 \text{ kN}$

stresses $\sigma_{x,Ed} = -55.4 \text{ N/mm}^2$, $\tau_{xz,Ed} = 0.0 \text{ N/mm}^2$

$|\sigma_{x,Ed}| = 55.4 \text{ N/mm}^2 < \sigma_{Rd} = 235.0 \text{ N/mm}^2 \Rightarrow U = 0.236 < 1$ **ok**

$|\tau_{xz,Ed}| = 0.0 \text{ N/mm}^2 < \tau_{Rd} = 135.7 \text{ N/mm}^2 \Rightarrow U = 0.000 < 1$ **ok**

$\sigma_v = 203.9 \text{ N/mm}^2 < \sigma_{Rd} = 235.0 \text{ N/mm}^2 \Rightarrow U = 0.867 < 1$ **ok**

Lk 2: $M_{y,Ed} = -351.0 \text{ kNm}$, $V_{z,Ed} = 75.0 \text{ kN}$

stresses $\sigma_{x,Ed} = -55.4 \text{ N/mm}^2$, $\tau_{xz,Ed} = 0.0 \text{ N/mm}^2$

$|\sigma_{x,Ed}| = 55.4 \text{ N/mm}^2 < \sigma_{Rd} = 235.0 \text{ N/mm}^2 \Rightarrow U = 0.236 < 1$ **ok**

$$|\tau_{xz,Ed}| = 20.1 \text{ N/mm}^2 < \tau_{Rd} = 135.7 \text{ N/mm}^2 \Rightarrow U = 0.148 < 1 \text{ ok}$$

$$\sigma_v = 235.2 \text{ N/mm}^2 > \sigma_{Rd} = 235.0 \text{ N/mm}^2 \Rightarrow U = 1.000 \leq 1 \text{ ok}$$

maximum utilization: $\max U_{ULS} = 1.000 \leq 1 \text{ ok}$

2.2. buckling of transverse loading (ULS)

slenderness $\lambda_F = (F_y/F_{cr})^{1/2} = 0.608$, $F_y = 987.5 \text{ kN}$

reduction factor $\chi_F = 0.822$

resistance of buckling $F_{z,Rd} = f_y \cdot L_{eff} \cdot t_w / \gamma_{M1} = 737.98 \text{ kN}$, $L_{eff} = \chi_F \cdot l_y = 406.4 \text{ mm}$, $l_y = 494.3 \text{ mm}$

verification: $F_{z,Ed}/F_{z,Rd} = 0.244 < 1 \text{ ok}$

interaction (without plated structures-/shear buckling):

transverse loading and equivalent stress $(\eta_2 + 0.8 \cdot \eta_1) / 1.4 = 0.746 < 1 \text{ ok}$

with $\eta_2 = F_{z,Ed}/F_{z,Rd} = 0.244$, $\eta_1 = \max U_{ULS} = 1.000$

3. final result

maximum utilization: $\max U = 1.000 \leq 1 \text{ ok}$

verification succeeded